

# Numerical Verification of Stainless Steel Overall Buckling Curves

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## Abstract

The introduction of new material grades or fabrication methods in engineering structures always raises concerns about the validity of current design rules, especially those based on earlier material or member testing. One of the fundamental structural checks is the overall stability of beams and columns. In this paper, a series of virtual buckling tests is calculated and compared to the Eurocode buckling curves. The main focus is on the applicability of ferritic steels that generally have different stress-strain behaviour than austenitic or duplex grades.

## Keywords

Stainless steel, buckling, finite element method, stress-strain relation, regression

## Highlights

Over 1,500 numerical calculations of models with shell elements, residual stress and nonlinear materials were carried out. Software for virtual testing in Abaqus was developed and used for this study. The recorded ultimate loads were recalculated to non-dimensional reduction factors. The results were compared to the standard Eurocode 3 buckling curves.

## 1 Introduction

Ferritic stainless steels are very competitive materials for load-bearing building structures, but they do not have such a long tradition as austenitic grades and a relatively low amount of experimental results from members is available to support design rules for such materials. The geometrically and materially nonlinear analysis on imperfect finite element models (GMNIA) has a great potential to simulate real experiments and, for instance, to computationally predict the buckling curve of a certain material and cross-section including the effects of fabrication such as enhanced strength and residual stress. The parametrical study was carried out to obtain parameters for overall buckling reduction calculation and was compared to the experimental tests [1-9] and Eurocode 3 buckling curves. Based on the review of available data [10], a preliminary numerical study [11,12] and experiments [6], several important parameters were recognised that have to be taken into account in this study.

### (a) Buckling modes

The study covers flexural buckling (FB) to the major and minor axes, torsional-flexural buckling (TFB) and lateral-torsional buckling of beams (LTB). These modes are also recognised in Eurocode 3, where the reduction factors are given separately for compressed columns and members subjected to bending.

### (b) Cross-sectional shapes and fabrication methods

Cold-formed open sections, hollow sections and welded open sections are the basic categories in Eurocode 3, Part 1-4. We selected cold-rolled (by circle-to-rectangle forming) hollow sections, press-braked open sections and welded I-sections in the present study because more detailed knowledge of the fabrication method was needed for accurate buckling strength prediction.

### (c) Material nonlinearity

The value of the Ramberg-Osgood coefficient  $n$  is generally higher in ferritic steels (black markers in Figure 1) but it can decrease during fabrication. It should be noted that a higher nonlinear factor usually leads to a higher buckling strength [11-13].

### (d) Strain hardening rate

The stress-strain relation beyond the yield point can be represented as the ratio of the ultimate stress  $\sigma_u$  or 1% proof stress  $\sigma_{1.0}$  and 0.2% proof stress  $\sigma_{0.2}$ . This parameter does not significantly affect the buckling strength, but the lower hardening rate also means lower strength enhancement in corners, which can lead to differences in the average member strength. We extrapolated the collected experimental stress-strain curves to 40% strain and used the corresponding stress  $\sigma_{40}$  in Figure 1 instead of the real measured ultimate stress because the value of ultimate strain  $\epsilon_u$  in the numerical models is always 40%. The lower limit of  $f_u/f_y$  ratio of stainless steels in Eurocode 3 is 1.1.

### (e) Design yield strength

The particular value of yield strength used in the member design does not significantly affect the buckling reduction factors, but it is essential to know to which of the following strengths it corresponds. Cold-formed profiles from sheets

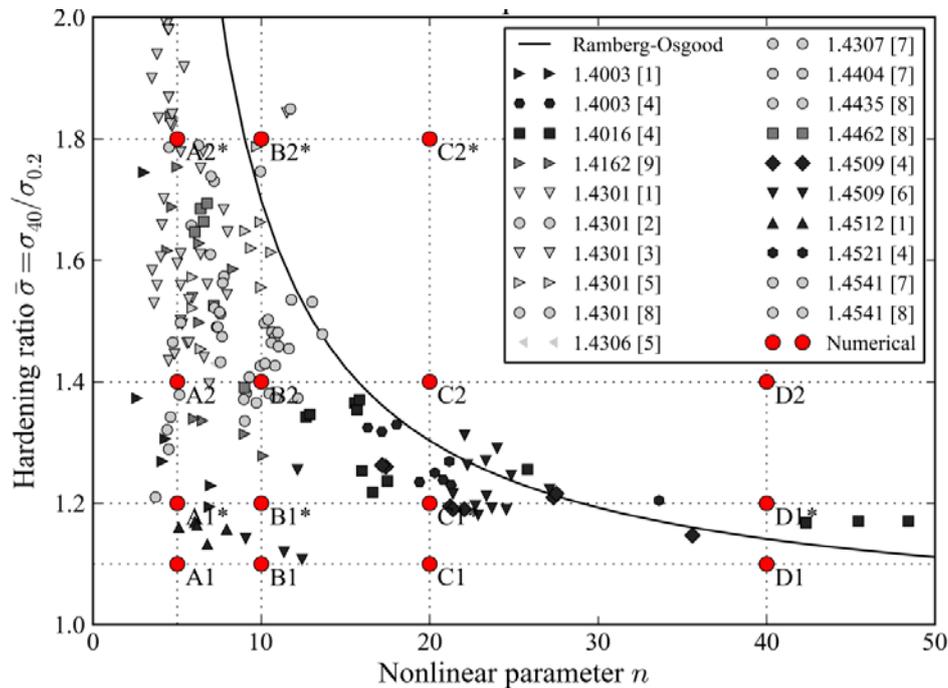
with virgin material strength  $\sigma_{0.2,v}$  may have higher values of yield point in the flat parts  $\sigma_{0.2,f}$  and corners  $\sigma_{0.2,c}$  due to the fabrication. Therefore, the average yield strength  $f_{y,a}$  can differ from the basic one  $f_{y,b}$  and would require a new set of buckling curves.

## 2 Materials and Cross-sections

The effect of material nonlinearity and work-hardening from the fabrication is described by two important material parameters: the nonlinear factor  $n$  and the ratio of ultimate and yield strength  $f_u/f_y = \sigma_u/\sigma_{0.2}$ . We used four groups of materials with different nonlinear factor  $n$  (groups A, B, C and D) varying from 5 to 40 and four levels of material hardening (groups 1, 1\*, 2 and 2\*) from 1.1 to 1.8. Their properties correspond to the basic materials in press-braked and welded sections and to the flat part materials in case of cold-rolled hollow sections. The initial modulus of elasticity  $E_0$  was always 200 GPa and the basic (flat parts) yield strength  $\sigma_{0.2}$  was selected as low as 250 MPa to produce conservative results. Stress-strain relationships are based on Mirambell and Real's two stage model [14], where the ultimate strain  $\varepsilon_u$  and the second nonlinear parameter  $m$  are 0.4 and 3 respectively in all cases. The 1.0% proof stress  $\sigma_{1.0}$ , the ultimate strength  $\sigma_u$  and the nonlinear parameter  $n$  are presented in Table 1.

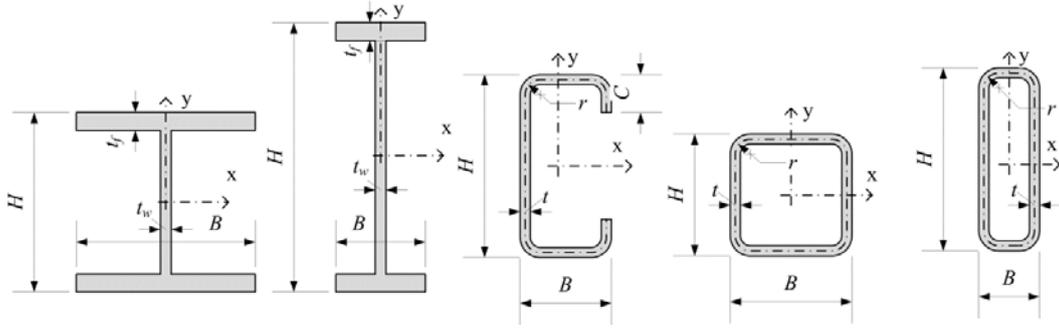
**Table 1 Variable material properties**

| Material | $\sigma_{1.0}$<br>(MPa) | $\sigma_u$<br>(MPa) | $n$ | Material | $\sigma_{1.0}$<br>(MPa) | $\sigma_u$<br>(MPa) | $n$ |
|----------|-------------------------|---------------------|-----|----------|-------------------------|---------------------|-----|
| A1       | 256.25                  | 275                 | 5   | A2       | 275                     | 350                 | 5   |
| B1       | 256.25                  | 275                 | 10  | B2       | 275                     | 350                 | 10  |
| C1       | 256.25                  | 275                 | 20  | C2       | 275                     | 350                 | 20  |
| D1       | 256.25                  | 275                 | 40  | D2       | 275                     | 350                 | 40  |
| A1*      | 262.5                   | 300                 | 5   | A2*      | 300                     | 450                 | 5   |
| B1*      | 262.5                   | 300                 | 10  | B2*      | 300                     | 450                 | 10  |
| C1*      | 262.5                   | 300                 | 20  | C2*      | 300                     | 450                 | 20  |
| D1*      | 262.5                   | 300                 | 40  | D2*      | 300                     | 450                 | 40  |



**Figure 1 Nonlinear factor and hardening ratio in collected material tests (black markers represent ferritic grades, light grey markers austenitic materials and dark grey markers indicate duplex steels)**

The study focuses on welded sections (I and H profiles), press-braked open sections (C-sections) and cold-rolled hollow sections (SHS and RHS). One cross-section was selected from each group with high  $A/W$  ratio to provide conservative results (see Figure 2 and Table 2). This ratio was always higher than in experiments reported in the commentary to the Design Manual for Structural Stainless Steel [15] and represented the higher boundary in the collected database of typical cross-sectional shapes. Sectional properties of welded I-sections were compared to the parameters of typical hot-rolled steel profiles (IPE and HE) and cold-formed cross-section geometrical properties are based on more than 800 cross-sectional shapes from EN 1019-2, SCI report [16] and Stalutube Oy data sheets [17]. Table 2 also shows additional important cross-sectional parameters such as the internal corner radius and thickness ratio  $r_i/t$  and the share of a cross-sectional area with enhanced material properties  $A_{enh}$ .



**Figure 2** Cross-section types used in the study

**Table 2** Cross-sectional parameters

| Section     | $H$<br>(mm) | $B$<br>(mm) | $C$<br>(mm) | $t_w$<br>(mm) | $t_f$<br>(mm) | $A/W_{el,x}$<br>( $\text{cm}^{-1}$ ) | $A/W_{el,y}$<br>( $\text{cm}^{-1}$ ) | $r_i/t$ | $A_{enh}/A$ |
|-------------|-------------|-------------|-------------|---------------|---------------|--------------------------------------|--------------------------------------|---------|-------------|
| I 150x50    | 150         | 50          | -           | 5             | 10            | 0.21                                 | 1.97                                 | -       | -           |
| H 100x100   | 100         | 100         | -           | 5             | 10            | 0.28                                 | 0.72                                 | -       | -           |
| C 72x36x4   | 72          | 36          | 15          |               | 4             | 0.47                                 | 1.30                                 | 1.0     | 24%         |
| RHS 72x24x4 | 72          | 24          | -           |               | 4             | 0.60                                 | 1.30                                 | 1.0     | 52%         |
| SHS 48x4    | 48          | 48          | -           |               | 4             | 0.69                                 | 0.69                                 | 1.0     | 62%         |

### 3 Finite element models

The numerical models were created and evaluated using the Abaqus plugin developed to simulate virtual testing of cold-formed steel members [18]. The functionality of the plugin was extended by script that automatically executed batches of pre-defined models with variable member lengths, evaluated results and plotted buckling curves.

#### 3.1 Enhanced material strength

The enhanced yield strength of press-braked open section C 72x36x4 was applied in the corners according to Cruise and Gardner's model [19] in Eq. (1).

$$\sigma_{0.2,c} = \frac{1.673 \sigma_{0.2}}{(r_i/t)^{0.126}} \quad (1)$$

The average yield strength  $f_{y,a}$  is therefore 294 MPa in the selected section where  $r_i/t = 1.0$  and  $A_{enh}/A = 24\%$ .

Cold-rolled hollow sections RHS 72x24x4 and SHS 48x4 have a larger area affected by the fabrication, which extends  $2t$  from the corners to the flat parts. The current predictive models for the enhanced strength in corners  $\sigma_{0.2,c}$  and flat faces  $\sigma_{0.2,f}$  of hollow sections are based on the knowledge of virgin material yield strength  $\sigma_{0.2,v}$ . Moreover, the model proposed by Cruise and Gardner [19] for hollow sections is limited to either low  $A/W$  or high  $f_u/f_y$  ratios, which was not acceptable in our study. Therefore we used Rossi's prediction [20] in Eq. (2) with the virgin material yield strength varying from 175 to 235 MPa that produced a flat part strength of exactly 250 MPa and a yield strength in the corners between 290 and 585 MPa. The calculation of parameters  $C_1$ ,  $C_2$  and  $\alpha$  is described in [20].

$$\sigma_{0.2,c} = \sigma_{0.2,v} + \frac{\sigma_{u,v}}{C_1 \left( \frac{r_i}{t/2} \right) + C_2 \left( \frac{r_i}{t/2} \right)^\alpha} \quad \text{and} \quad \sigma_{0.2,f} = \sigma_{0.2,v} + \frac{\sigma_{u,v}}{C_1 \left[ \frac{(b+d)/\pi}{t/2} \right] + C_2 \left[ \frac{(b+d)/\pi}{t/2} \right]^\alpha} \quad (2)$$

#### 3.2 Residual stresses

Membrane stresses from welding were considered in the I and H sections. The model for carbon steels from the Swedish standard BSK 99 that was proposed by Gardner and Cruise [21] for ferritic fabricated sections was used, where the compressive stress  $\sigma_c$  is calculated to maintain the equilibrium (Eq. (3)). The compressive stress was therefore 92.5 and 72.6 MPa in I and H sections respectively.

$$\sigma_c = \frac{2.25(t_f + t_w)}{B + 0.5H - 2.25(t_f + t_w)} \sigma_{0.2} \quad (3)$$

Bending residual stresses of 36% of basic yield strength  $f_{y,b}$  were inserted in corner areas and 15% of  $f_{y,b}$  in flats according to Gardner and Cruise [21] in press-braked open sections. In the case of hollow sections, the highest level of

bending residual stresses in corners and flat parts was 37% and 63% respectively. Membrane stresses were neglected in both types of cold-formed sections according to [21].

### 3.3 Initial imperfections

All finite element models were perturbed prior to the arc-length nonlinear calculation. The initial imperfections were distributed according to the first overall buckling shape with positive critical load produced by the linear eigenvalue analysis (LEA). The search for such shape can be unsuccessful due to very low member slenderness; therefore we used constrained models according to the study by Zhang and Tong [22], which also provided results for members where local or distortional buckling dominated.

The fabrication tolerance  $L/750$  was used as the imperfection amplitude in the preliminary study [11]. Since residual stresses are included in FE models this time, lower amplitude was needed that represents only the pure geometrical imperfections. We selected  $L/1500$ , which is widely used in similar studies [23,24].

## 4 Methods

Numerical models of full-section tension tests with enhanced material properties and residual stresses were calculated before the buckling curves were evaluated, in order to verify the predicted average yield strength  $f_{y,a}$  and reduced nonlinear factor  $n$ . Changes in material nonlinearity due to cold-forming would affect the estimated impact of the original  $n$  factor on the buckling curve. In compression or bending tests simulations, ten points of each buckling curve were calculated with the nondimensional slenderness  $\lambda$  ranging from 0.18 to 4.0. The nonlinear regression analysis provided parameters  $\alpha$  and  $\lambda_0$  for the Ayrton-Perry curve [25] fitted to the calculated points (see Figure 3).

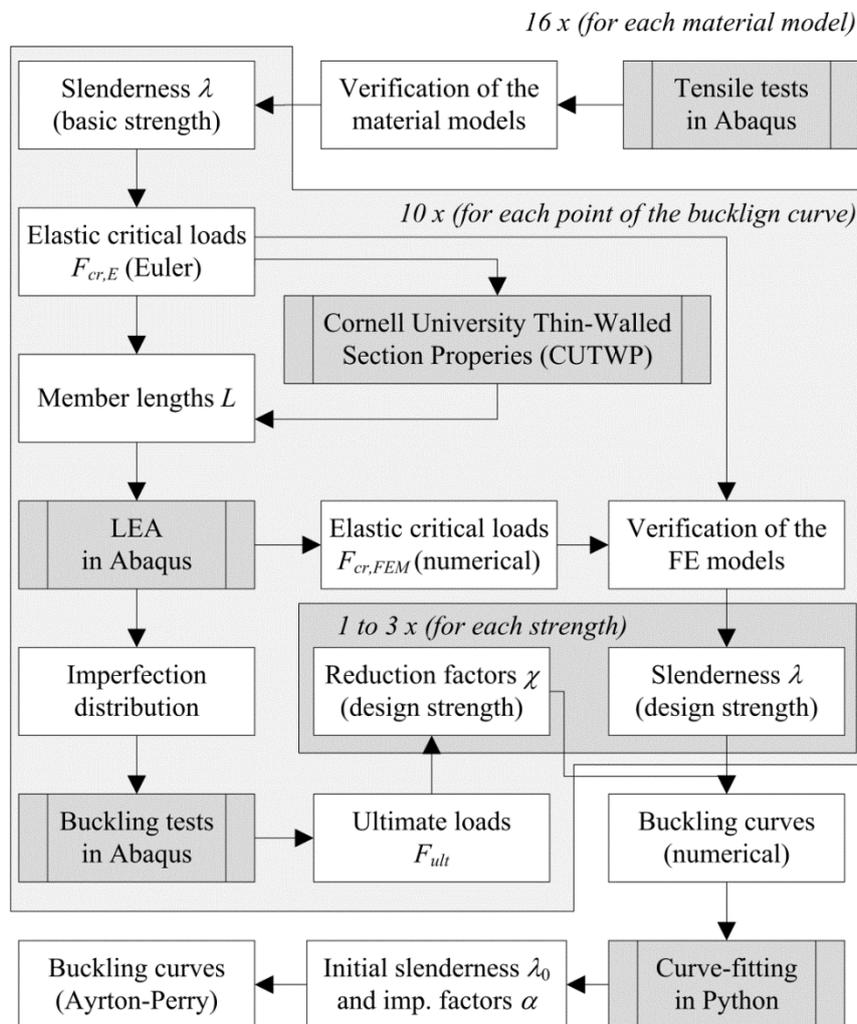


Figure 3 Numerical verification of Eurocode buckling curves

### (a) Tensile tests

One metre long members with no initial geometrical imperfections were loaded in tension and the load-displacement relation was recorded. The stress-strain curve obtained from the numerical results was compared to the original constitutive model. The least square optimisation tool described in Appendix E of VTT's report [6] generated the real

yield strength and nonlinear parameter  $n$  of the whole cross-section. The calculated strength was then compared to that predicted by Rossi's model [20].

### (b) Nondimensional slenderness $\lambda_{f(b)}$

Ten values of nondimensional slenderness were selected from 0.18 to 4.00 in the sequence  $\lambda_{f(b)} = \sqrt{2}^N$  for  $N$  from -5 to 4. They are assumed to be valid in combination with the basic (or flat parts) strength 250 MPa.

### (c) Elastic critical loads $F_{cr,E}$ ( $M_{cr,E}$ )

The theory of elastic buckling was used to predict the value of the critical load for compressed members and members subjected to bending according to Eq. (4) where  $A$  and  $W$  were the cross-sectional area and section modulus respectively.

$$F_{cr,E} = \frac{f_{y,f(b)} A}{\lambda_{f(b)}^2} \text{ and } M_{cr,E} = \frac{f_{y,f(b)} W}{\lambda_{f(b)}^2} \quad (4)$$

### (d) Member lengths $L$

The length of the member was calculated directly in simple flexural buckling cases (Eq. (5)) or with CUTWP software [26] in more complex torsional buckling cases using the iterative approach.

$$L = \lambda_{f(b)} \cdot \pi i \sqrt{E / f_{y,f(b)}} = \pi \sqrt{EI / F_{cr,E}} \quad (5)$$

### (e) Elastic critical loads $F_{cr,FEM}$ ( $M_{cr,FEM}$ )

The critical load from the linear eigenvalue analysis (LEA) of the model with non-uniform distribution of material properties and residual stresses does not necessarily have to be the same as the predicted one. Both results were compared, and when their difference was higher than 5%, the basic nondimensional slenderness was corrected using Eq. (6).

### (f) Nondimensional slenderness $\lambda_v$ , $\lambda_{f(b)}$ and $\lambda_a$

Each type of yield strength other than the basic one (virgin or average) which can be used in design would require different buckling curves to be plotted. Therefore the values of nondimensional slenderness were corrected accordingly using Eq. (6) where  $f_y$  stands for the virgin material strength  $f_{y,v}$ , basic (flat parts) strength  $f_{y,f(b)}$  or the average strength  $f_{y,a}$ .

$$\lambda = \sqrt{f_y A / F_{cr,FEM}} \text{ and } \lambda = \sqrt{f_y W / M_{cr,FEM}} \quad (6)$$

### (g) Imperfection distribution

The shapes of initial imperfections were extracted from the results of LEA analysis and amplified to match selected amplitude  $L/1500$  in the node with the highest displacement.

### (h) Ultimate loads $F_{ult}$ ( $M_{ult}$ )

The peak loads in arc-length Riks analysis [27] were recorded for each evaluated case. This is the common approach in geometrically and materially nonlinear analysis of imperfect models (GMNIA).

### (i) Nondimensional reduction factors $\chi$

The value of the reduction factor has to correspond to the yield strength that is used in the design or higher to provide safe results. Therefore it was possible to create three sets of reduction factor-to-slenderness plots for  $f_{y,a}$  (average strength),  $f_{y,f}$  (flat faces) and virgin material strength  $f_{y,v}$  of cold-rolled hollow sections according to Eq. (7). Press-braked open sections yielded two sets of buckling curves with the average and basic strength  $f_{y,b}$ , and only one basic strength was needed for the welded open sections.

$$\chi = F_{ult} / f_y A \text{ and } \chi = M_{ult} / f_y W \quad (7)$$

### (j) Buckling curves

Nondimensional reduction factors were plotted against nondimensional slenderness and formed the approximation of 208 buckling curves (48 for welded open sections, 64 for press-braked open sections and 96 for cold-rolled hollow sections). The same least square optimisation method as described in the tensile tests was applied to the calculated sets of nondimensional parameters. Finally, the values of the buckling curve parameters were selected to match the current Eurocode buckling curves, where the initial slenderness is either 0.2 or 0.4, and the imperfection factor is 0.21, 0.34, 0.49 or 0.76.

## 5 Results and Discussion

### 5.1 Tensile tests

The nonlinearity of axially-loaded member load-displacement relation increases during cold forming; therefore we expected a lower nonlinear factor for the whole cross-section than in the original material. This effect can be verified by a simple tensile test of the whole cross-section. Using the finite element method, we were able to estimate how much the nonlinearity was affected by enhanced material properties and how big the effect of residual stress was. Table 3 shows curve-fitted results of an average material yield strength  $\sigma_{0.2,TT}$  and nonlinear factor  $n_{TT}$  from full-section tensile tests with and without residual stresses.

**Table 3** Curve-fitted parameters of stress-strain curves from the full-section tests

| Cross-section | Material | with residual stress |                         |          | without residual stress |          |
|---------------|----------|----------------------|-------------------------|----------|-------------------------|----------|
|               |          | $n$                  | $\sigma_{0.2,TT}$ (MPa) | $n_{TT}$ | $\sigma_{0.2,TT}$ (MPa) | $n_{TT}$ |
| C 72x34x4     | A2       | 5                    | 294.3                   | 4.63     | 294.9                   | 4.72     |
|               | B2       | 10                   | 296.3                   | 8.04     | 296.7                   | 8.58     |
|               | C2       | 20                   | 297.3                   | 12.73    | 297.6                   | 15.03    |
| SHS 48x4      | A2       | 5                    | 275.4                   | 4.96     | 276.4                   | 4.94     |
|               | B2       | 10                   | 275.5                   | 7.49     | 276.5                   | 9.63     |
|               | C2       | 20                   | 275.7                   | 12.02    | 276.3                   | 17.89    |
| RHS 72x24x4   | A2       | 5                    | 272.5                   | 4.99     | 273.0                   | 4.95     |
|               | B2       | 10                   | 271.7                   | 7.59     | 272.7                   | 9.60     |
|               | C2       | 20                   | 271.8                   | 12.14    | 272.4                   | 17.99    |

As can be seen in Table 3, differences between nonlinear factors decreased partly because of residual stress and partly due to the non-uniform material distribution. Smaller effect of  $n$  factor than in the preliminary study [11,12], where the fabrication was not considered, is then expected especially in hollow sections.

### 5.2 Buckling tests

Calculated lengths of columns and beams are presented in Table 4. These members were loaded with concentric axial force or end-moments according to their expected failure mode. Pinned-pinned supports and concentric axial loading were applied in the flexural buckling study. The single-symmetric members were forced to fail in torsional-flexural buckling, fixing both ends in minor axis bending, torsion and warping. In the case of lateral-torsional buckling, members were simply supported and loaded with end-moments. Both ends were additionally restrained against torsion and warping. Table 5 shows the ultimate loads of GMNIA calculations.

**Table 4** Lengths of tested members in mm

| $\lambda_{f(b)}$ | H 100x100 |         | I 150x50 |         | C 72x34x4 |         | SHS 48x4 | RHS 72x24x4 |
|------------------|-----------|---------|----------|---------|-----------|---------|----------|-------------|
|                  | FB maj.   | FB min. | LTB      | FB min. | TFB       | FB maj. | FB min.  |             |
| 0.18             | 663       | 174     | 301      | 196     | 282       | 292     | 151      |             |
| 0.25             | 938       | 247     | 430      | 278     | 401       | 413     | 213      |             |
| 0.35             | 1327      | 349     | 618      | 393     | 576       | 584     | 302      |             |
| 0.50             | 1876      | 493     | 904      | 555     | 838       | 826     | 427      |             |
| 0.71             | 2653      | 697     | 1365     | 785     | 1251      | 1168    | 603      |             |
| 1.00             | 3752      | 986     | 2183     | 1111    | 1923      | 1652    | 853      |             |
| 1.40             | 5307      | 1395    | 3802     | 1570    | 2981      | 2336    | 1207     |             |
| 2.00             | 7505      | 1973    | 7170     | 2221    | 4477      | 3304    | 1706     |             |
| 2.80             | 10613     | 2790    | 14066    | 3141    | 6536      | 4672    | 2413     |             |
| 4.00             | 15010     | 3945    | 28000    | 4442    | 9390      | 6607    | 3413     |             |

**Table 5 Minimum ultimate loads in kN in compression and kNm in bending tests**

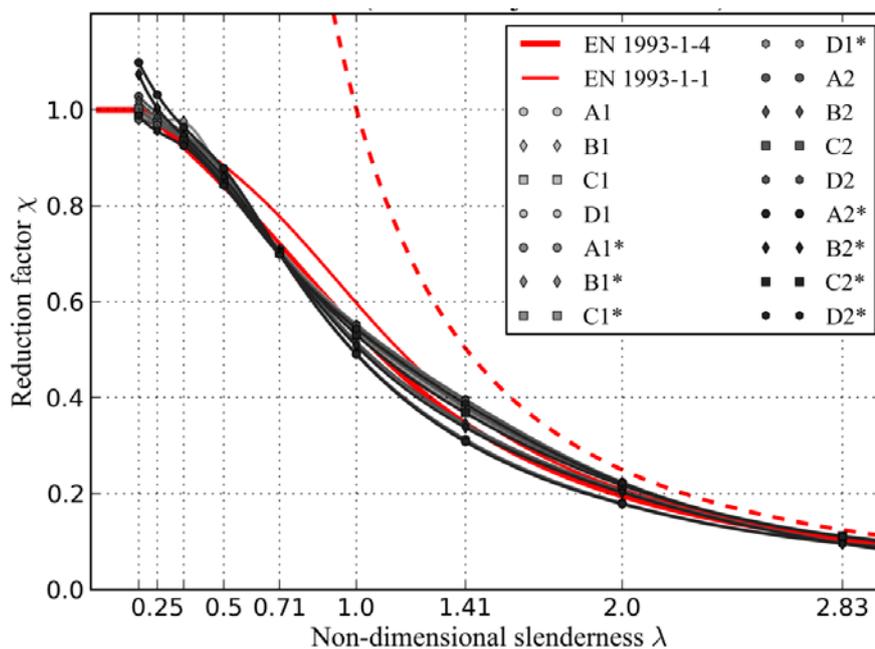
| $\lambda_{f(b)}$ | H 100x100 | I 150x50 |       | C 72x34x4 |       | SHS 48x4 | RHS 72x24x4 |
|------------------|-----------|----------|-------|-----------|-------|----------|-------------|
|                  | FB maj.   | FB min.  | LTB   | FB min.   | TFB   | FB maj.  | FB min.     |
| 0.18             | 599.7     | 422.4    | 23.54 | 166.6     | 172.9 | 175.6    | 175.1       |
| 0.25             | 593.4     | 411.8    | 23.01 | 162.5     | 170.7 | 163.0    | 163.8       |
| 0.35             | 574.8     | 390.6    | 22.12 | 155.3     | 165.8 | 154.1    | 152.1       |
| 0.50             | 522.2     | 344.1    | 20.25 | 142.8     | 154.6 | 141.4    | 142.1       |
| 0.71             | 428.6     | 269.2    | 16.74 | 120.7     | 130.9 | 119.8    | 124.1       |
| 1.00             | 300.9     | 184.9    | 12.39 | 89.9      | 96.4  | 87.9     | 91.1        |
| 1.40             | 189.8     | 114.4    | 8.00  | 57.6      | 59.5  | 52.4     | 54.9        |
| 2.00             | 109.7     | 65.9     | 4.41  | 32.5      | 32.4  | 28.5     | 28.8        |
| 2.80             | 59.3      | 35.9     | 2.27  | 17.0      | 16.7  | 14.5     | 14.4        |
| 4.00             | 30.9      | 19.0     | n/a   | 8.7       | 8.5   | 5.4      | 5.5         |

**5.3 Reduction factors**

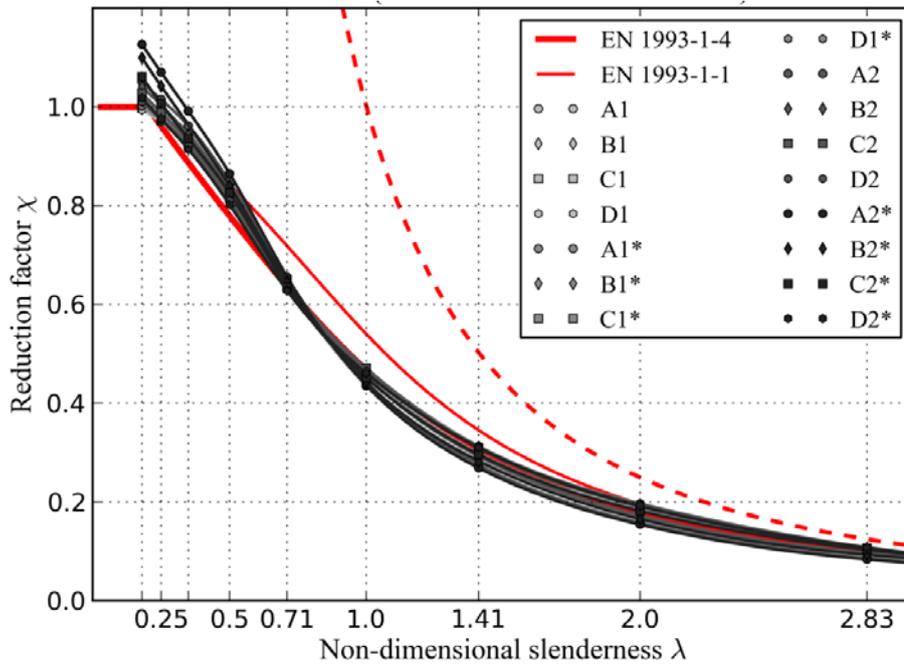
Nondimensional reduction factors were calculated for all materials and all types of yield strength. Here, we present the most important results with the basic or flat part strengths that are currently used in the Eurocode 3, Part 1-4. Figures 4 to 10 show the calculated numerical results for all materials and basic or flat part strength. The effect of using virgin material strength of cold-rolled hollow sections or using average strength of press-braked and cold-rolled sections is briefly described in this chapter. Presented curves are overlapping in some cases due to very small differences between reduction factors of welded and press-braked sections with materials with the same  $f_u/f_y$  ratio. Numerical results are compared to the Eurocode 3 buckling curves for stainless steels (EN 1993-1-4) and carbon steels (EN 1993-1-1 and EN 1993-1-3) according to Table 6.

**Table 6 Imperfection factors  $\alpha$  and initial slenderness  $\lambda_0$  of Eurocode 3 buckling curves**

|   | H 100x100 | I 150x50 |      | C 72x34x4 |      | SHS 48x4 | RHS 72x24x4 |
|---|-----------|----------|------|-----------|------|----------|-------------|
|   | FB maj.   | FB min.  | LTB  | FB min.   | TFB  | FB maj.  | FB min.     |
| EN 1993-1-4: Supplementary rules for stainless steels                 |           |          |      |           |      |          |             |
| $\alpha$  | 0.49      | 0.76     | 0.76 | 0.49      | 0.34 | 0.49     | 0.49        |
| $\lambda_0$   | 0.2       | 0.2      | 0.4  | 0.4       | 0.2  | 0.4      | 0.4         |
| EN 1993-1-1: General rules and rules for buildings                    |           |          |      |           |      |          |             |
| EN 1993-1-3: Supplementary rules for cold formed members and sheeting |           |          |      |           |      |          |             |
| $\alpha$  | 0.34      | 0.49     | 0.76 | 0.34      | 0.34 | 0.49     | 0.49        |
| $\lambda_0$   | 0.2       | 0.2      | 0.2  | 0.2       | 0.2  | 0.2      | 0.2         |

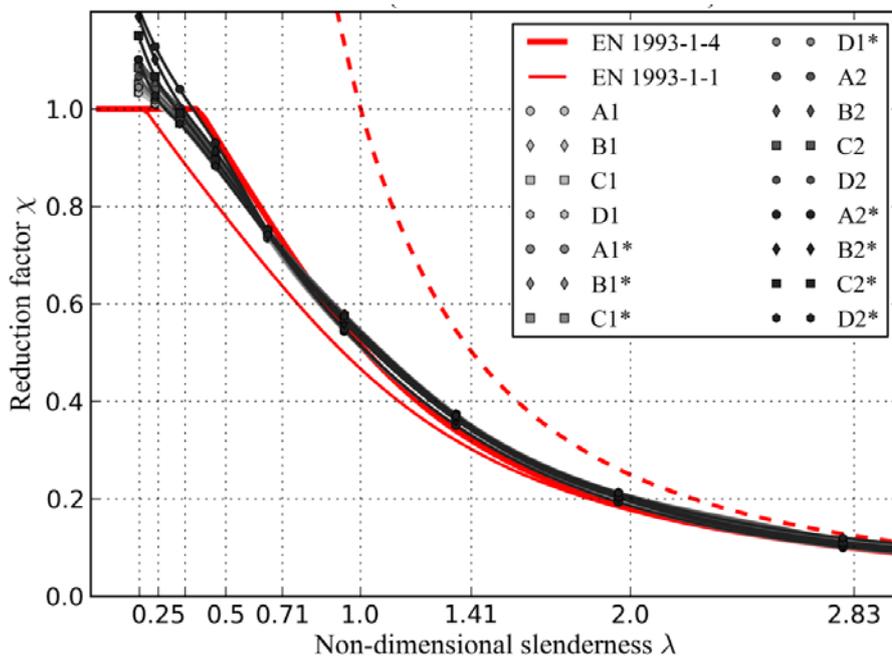


**Figure 4 H 100x100 flexural buckling - major axis**



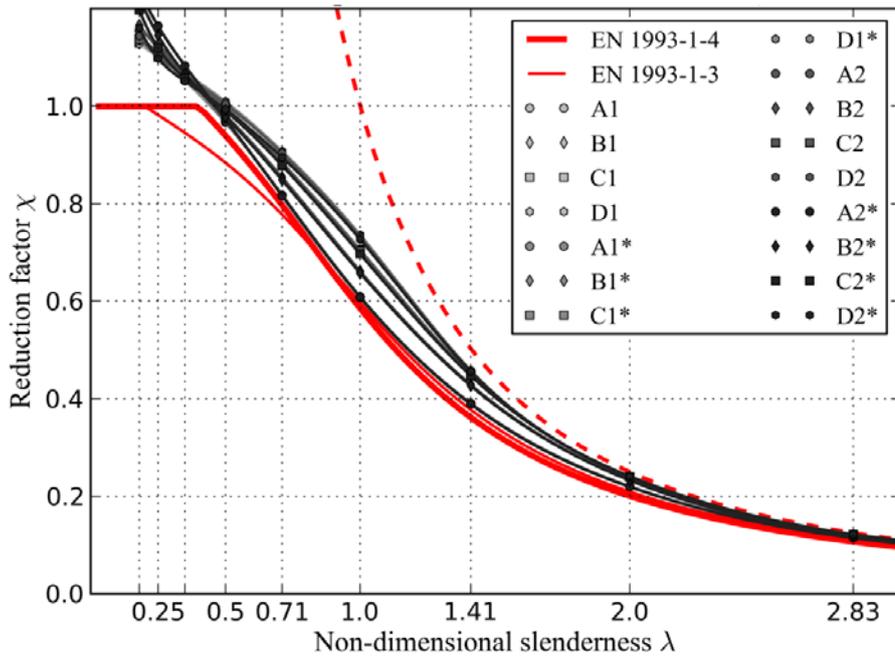
**Figure 5 I** 150x50 flexural buckling - minor axis

Calculated reduction factors of welded open sections indicate that the flexural buckling to the major or minor axis (Figures 4 and 5) corresponds to the EN 1993-1-4 buckling curves. Higher  $n$  values may yield in a slight improvement in buckling resistance for higher slenderness levels than 0.71.

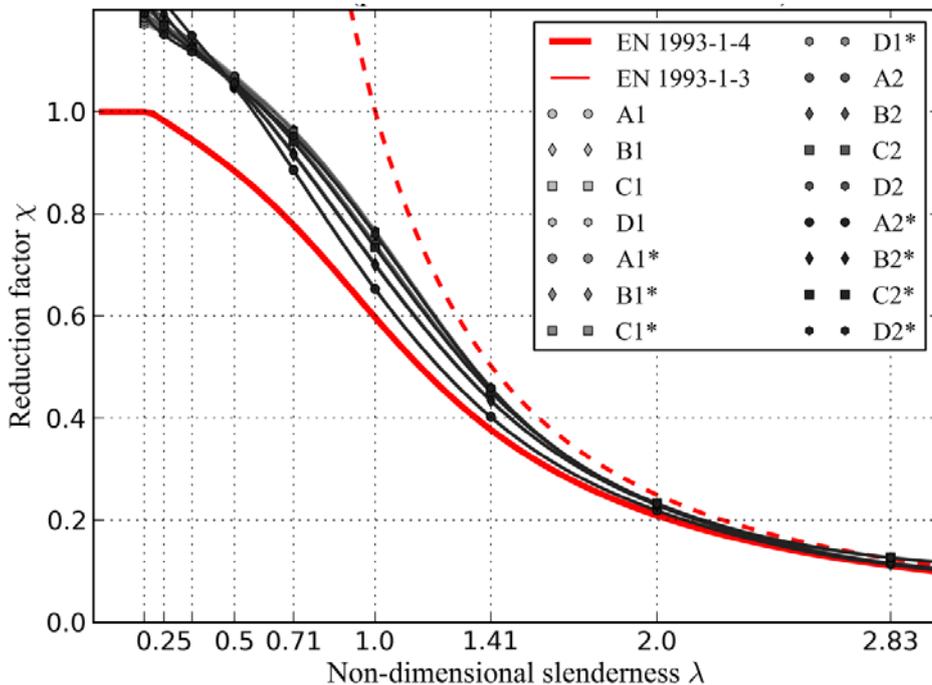


**Figure 6 I** 150x50 lateral-torsional buckling

Lateral-torsional buckling tests resulted in good correlation to the EN 1993-1-4 when the plastic section modulus was used, and therefore are conservative in elastic design.



**Figure 7** C 72x34x4 flexural buckling



**Figure 8** C 72x34x4 torsional-flexural buckling

Basic strength can be used together with the EN 1993-1-4 buckling curve in case of flexural and torsional-flexural buckling of the press-braked C-sections (Figures 7 and 8), but the use of average strength provides unsafe results in the case of flexural buckling. For instance, the slenderness 1.0 with the average strength 250 MPa will change to 1.08 for the average strength 294 MPa, but the reduction factor in flexural buckling with material A1 will drop from 0.61 to 0.52, which is lower than the EN 1993-1-4 design curve, where the reduction factor is 0.55. Results are not sensitive to the strain hardening rate, however significant improvement of predicted strength would be possible by increasing the nonlinear factor  $n$ . The results of the torsional-flexural buckling strength were rather conservative, but this could also be caused by the selection of the cross-sectional shape which was based on compression and bending resistance (particularly the  $A/W$  ratio).

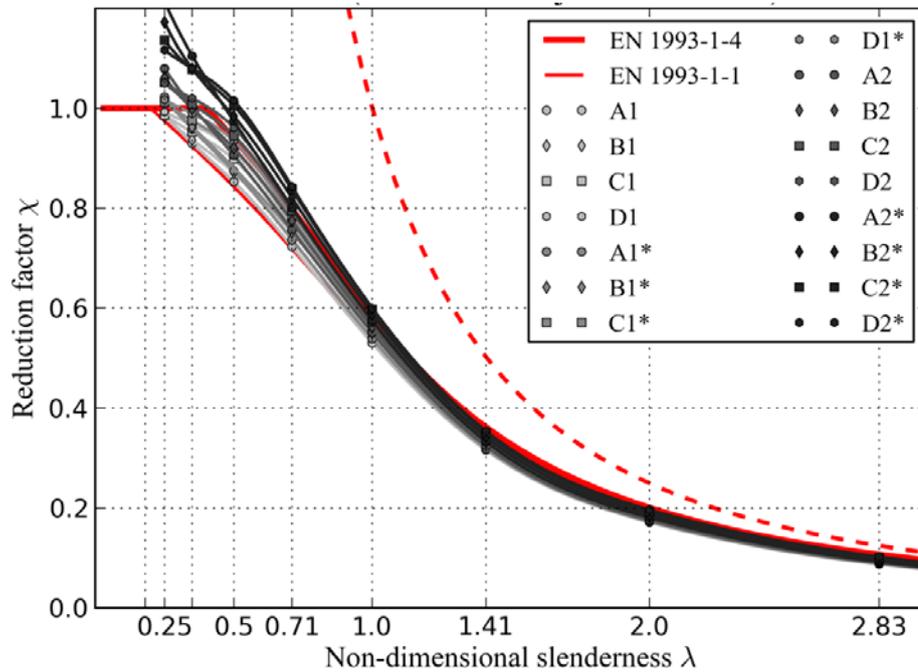


Figure 9 SHS 48x4 flexural buckling – major axis

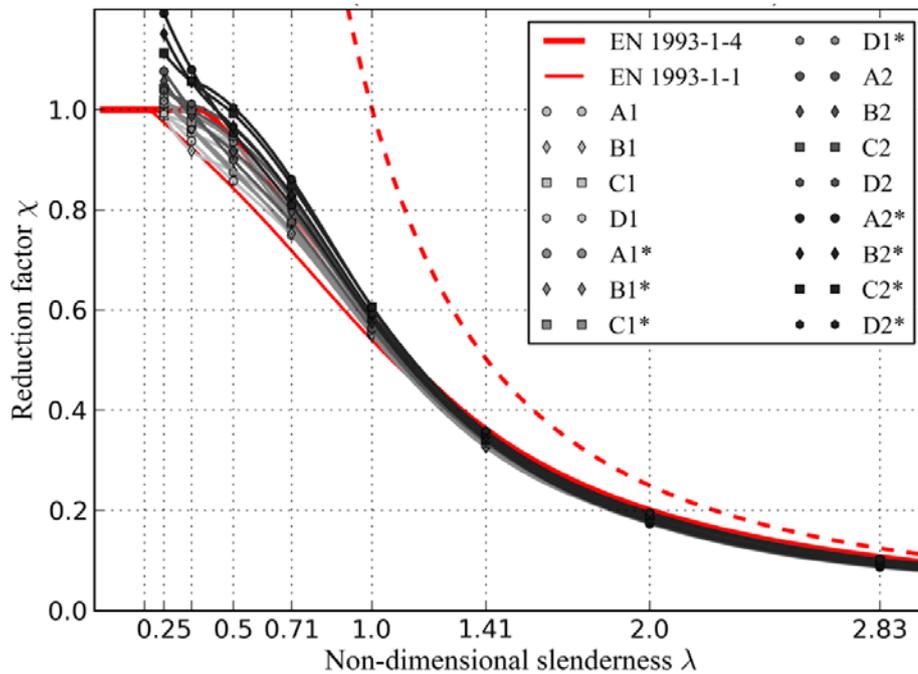


Figure 10 RHS 72x24x4 flexural buckling – minor axis

The use of virgin material strength always provides conservative results in cold-rolled hollow sections especially at lower slenderness. A lower buckling strength prediction than the EN 1993-1-4 buckling curve was achieved using material models with a high strain hardening ratio (A1-D1 and A1\*-D1\*) and the flat parts strength (Figures 9 and 10). However, the results with an average strength are not sensitive to this ratio and they are always unsafe.

## 6 Conclusions

Welded profiles were tested in major and minor axis flexural buckling and lateral-torsional buckling. The material model was uniform in the whole cross-section with the single material yield strength  $f_{y,b}$ . The variation of strain hardening ratio  $f_u/f_y$  did not significantly affect the results and the effect of the nonlinear parameter  $n$  was also relatively small. Calculated results indicate that the current buckling curves for flexural buckling of welded open sections can be used with all materials.

Press-braked channels were tested in minor axis flexural and torsional-flexural buckling. The basic yield strength  $f_{y,b}$  and the average strength  $f_{y,a}$  were used in evaluation. The use of average strength would require the recalibration of the buckling curve in Eurocode, which is only valid in combination with the basic strength. Designers may also benefit from a higher nonlinear parameter  $n$ , which is typical in ferritic stainless steels.

Cold-rolled hollow sections were tested in minor and major axis flexural buckling. Virgin material strength  $f_{y,v}$ , flat parts strength  $f_{y,f}$  and average strength  $f_{y,a}$  were used. While the application of virgin strength resulted in conservative results, average strength would require new buckling curves as in the case of press-braked sections. The situation was not so clear with flat parts strength, which is nowadays used in the design code. The calculations were very sensitive to the strain hardening ratio due to the differences in enhanced material yield strength prediction, and models containing materials such as ferritic steels with a lower  $f_u/f_y$  ratio produced lower strength than the Eurocode buckling curve, especially in lower slenderness ranges. Using the carbon steel buckling curve with the initial slenderness 0.2 would be more appropriate in such cases.

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